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HOUSTON ASTRONAUTICS DIVISION

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SPACE SHUTTLE ENGINEERING AND OPERATIONS SUPPORT

1.3-DN-C0201-C02

TRACKING SUBSYSTEM TEST REQUIREMENTS SURVEY

AVIONICS SYSTEM ENGINEERING

17 MARCH 1975

This Design Note is Submitted to NASA Under Task Order
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1.0 SUMMARY

A survey of the test and checkout requirements of the tracking portion of the Communications and Tracking subsystem has been performed to evaluate adequacy of planned tests and test requirement documents. Emphasis is placed on identifying test completeness, duplications and omissions. Items that may save time, aid in testing and present a more complete integrated test program are also noted. Table I summarizes the results of this survey with more detailed discussion included in Section 3 and Table III. It is planned to continue the survey by covering other communications and tracking functions (command and telemetry) in subsequent reports.

TABLE I

Summary of Test Requirements Survey

DISCREPANCY	RECOMMENDATIONS
<ul style="list-style-type: none"> • The SAIL Integrated Test Plan, Vol. II (Ref. Q) does not specifically include testing for verification of redundancy management • The SAIL and ESTL "Orbital Tracking" tests requirements are nearly the same. • No "Rendezvous Data Acquisition and Convergence Times" tests are called out for OFT (Flight). • OFT (Ground) testing of Rendezvous Radar/Comm A is too detailed 	<ul style="list-style-type: none"> • Tests should be added to accomplish verification of redundancy management. • Perform a combined SAIL/ESTL test to reduce testing. • Add "Rendezvous Data Acquisition and Convergence Times" tests for OFT (Flight) test. • Perform only self test and system functional type tests. Delete detailed performance tests such as heater temperature and RF spectrum measurements.

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TABLE I (Continued)

DISCREPANCY	RECOMMENDATIONS
<ul style="list-style-type: none">● Power output measurements (cut of antenna) are not made for the TACAN, Radar Altimeter, MSBLS, and S-Band transmitter subsystems during ALT ground test.● Many of the ALT ground tests are too detailed.● The C-Band beacon (OV 101 only) is not referenced in either the VEI or the MVP.	<ul style="list-style-type: none">● Recommend that these measurements be taken at the output of the antenna coupler system rather than at the black box.● ALT ground testing should eliminate detailed testing such as coding rejection limits, AGC slew rates, spectrum check, pulse width, and repetition rates.● Update VEI and MVP as required to reflect use of C-Band beacon on OV 101.

2.0 INTRODUCTION

The purpose of this report is to identify the planned tracking system tests to be performed at each major test facility and determine if these tests and locations satisfy Orbiter Vehicle End Item Specification (VEI) (Reference A) and the Shuttle Master Verification Plan (MVP) (Reference B) requirements.

Program requirements that are to be controlled by the NASA Space Shuttle Program Manager (Level II) are documented within the volumes of JSC 07700, Space Shuttle Program Definition and Requirements. Volume X of JSC 07700 defines the Space Shuttle Flight and Ground system requirements to be used by all NASA and contractor organizations involved in the design, development, production, test and operation of the Space Shuttle flight and ground systems. Volume X of JSC 07700 (Reference P) also defines the Space Shuttle MVP as the source of verification program requirements to which all elements of the Space

Shuttle Program must adhere. The element and combined Test Requirement Specification Documents (TRSD) and test plans are developed from the various volumes of the MVP and VEI requirements documents. Figure 1 presents a description of document hierarchy and the release status of those documents used in this survey. The documents used in this survey are also shown in the list of reference documents in Section 5.0.

3.0 DISCUSSION

3.1 General

To initiate the survey, two comparisons were made. First, the MVP (Vol. III), was compared with the VEI specification to determine if all VEI requirements were being adequately verified. It was found that the MVP specifies test locations for each VEI tracking requirement. The second comparison made in this survey involved the tests planned for each major test facility. These tests were compared with each other and with the VEI requirements to identify test duplications, omissions, and test document discrepancies.

The major facilities addressed in this design note are as follows:

- a) Avionics Development Lab (ADL) at Rockwell, Downey
- b) Shuttle Avionics Integration Lab (SAIL) at JSC
- c) Electronics System Test Lab (ESTL) at JSC
- d) SAIL/ESTL at JSC
- e) In Process Tests at Rockwell, Palmdale
- f) Approach and Landing Tests (ALT), both Ground and Flight at Edwards AFB
- g) Orbital Flight Tests, both Ground and Flight at KSC

There are five tracking requirement paragraphs reflected in the VEI. These requirements are: Rendezvous Tracking, Rendezvous Data Acquisition and Convergence Times, Atmospheric Flight Support Tracking, Final Approach and

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Landing Tracking, and Orbital Tracking. Each have unique functional requirements associated with them which will be discussed in detail in this design note.

Table II is an overview showing the various test locations where testing is to be performed versus the applicable VEI functional requirement to be verified.

Table III is a detailed listing of the planned tests to be performed at each test facility; the MVP designated verification program; and the related VEI requirement. Test duplications, omissions and comments are denoted where applicable. Test plans and test requirements documents used in this comparison are shown in the list of reference documents.

3.2 Rendezvous Tracking

This design note section includes discussion and comments on the adequacy of the rendezvous tracking tests. The subcontractor test article and OFT-OV3 are planned for the official verification programs of the rendezvous tracking test requirements. Comments are to be found below and in the remarks section of Table III.

3.2.1 Palmdale, OV Ground Test

In general, the Palmdale OV Ground Tests for rendezvous tracking demonstrate system rates, accuracies and capabilities. The Requirement/Parameter Description for the Radar Long Range Mode is listed as TBD in Reference F. This requirement should be determined to insure adequate testing of the Radar Long Range Mode.

3.2.2 ALT (Ground)

The ALT Ground test document (Reference G) references a test of the Ku-Band rendezvous tracking radar. This reference should be deleted

since no rendezvous radar is included on OV-101.

3.2.3 OFT (Ground)

The KSC ground testing for the rendezvous radar system is too detailed. Testing should be reduced to eliminate items such as testing black box specifications. It is recommended that detailed performance tests such as heater temperature and RF spectrum measurements be deleted. Only self tests and system functional type tests should be performed.

3.3 Rendezvous Data Acquisition and Convergence Times

The subcontractor test article is designated by the MVP for verification of these requirements. Major test facility testing is quite limited as the SAIL is the only apparent test facility to perform testing of Rendezvous Data Acquisition and Convergence Times.

3.3.1 OFT (Flight)

It is recommended that Rendezvous Data Acquisition and Convergence Times testing be added to the OFT Flight test program.

3.4 Atmospheric Flight Support Tracking

The subcontractor test article is designated by the MVP as the verification program for Atmospheric Flight Support Tracking.

C-Band beacon functional verification tests (for OV 101 only) are called out in Reference F; however, there are no references to the C-Band beacon in either the VEI Specification (Reference A) or the MVP (Reference B). The VEI and MVP should be updated to reflect C-Band beacon use during ALT.

3.4.1 ADL

The ADL test document (Reference M) indicates that development and trend testing at the TACAN operational range will include power output for the air-to-air mode. However, the air-to-air mode will not be implemented. It is recommended that air-to-air mode testing not be performed.

3.4.2 SAIL

The SAIL test document (Reference J) currently lists no tests to be performed relating to Atmospheric Flight Support Tracking. It is recommended that SAIL perform testing to verify systems operations of a TACAN subsystem with other subsystems. Tests should be added to accomplish verification of redundancy management.

3.4.3 Palmdale, OV Ground Test

The test method for the TACAN power output T/R mode is listed as TBD in Reference F. The C-Band Beacon functional verification tests are also listed as TBD. It is recommended that test methods for each be determined in order to insure completion of TACAN and C-Band beacon testing in a timely manner.

3.4.4 ALT (Ground)

TACAN sensitivity measurements should take into account the latest test philosophy of using antenna couplers rather than a hardline directly to the hardware. Sensitivity measurement specifications as called out in Reference G must be changed because of additional power losses due to the use of antenna couplers. Many of the test requirements listed for ALT ground testing are duplicates or more detailed than those for Palmdale in process testing. It is therefore recommended that performance checks such as memory check, range and bearing limits, and adjacent channel rejection be deleted.

3.4.5 OFT (Ground)

In general the KSC ground testing is too detailed. It is recommended that testing be limited to self tests and system functional type tests.

3.5 Final Approach and Landing Tracking

The verification points for Final Approach and Landing Tracking are the SAIL and the OV-101 ALT as specified by the MVP.

3.5.1 SAIL

The SAIL should utilize flight type hardware for system verification testing of NAVAIDS rather than evaluating MSBLS and RA by using math models since SAIL is designated as a verification point. Reference K should delete references specifying using math models of landing aid RF equipment for verification of these systems. Tests should be added to accomplish verification of redundancy management.

3.5.2 ALT (Ground)

Some of the ground operations test requirements at Edwards are too detailed as called out in Reference R. For example testing parameters such as coding rejection limits, AGC slew rate, spectrum checks, pulse width, repetition rates, and interpulse spacing (of the MSBLS DME pulses) is not considered necessary. Power output measurements for the RA and MSBLS should be made out of the antenna in order to follow the latest testing philosophy of using antenna couplers rather than hardlining directly to the hardware.

3.5.3 OFT (Ground)

In general the KSC test requirements are too detailed. Tests such as DME pulse width and repetition rate, spectrum checks and AGC slew rate checks should be deleted.

3.6 Orbital Tracking

The ESTL and OV2 flight test are the MVP verification points for Orbital Tracking. The tests called out in the test documents should adequately satisfy the VEI requirements.

3.6.1 ESTL

The test requirements for ESTL are nearly the same as the test requirements for SAIL. Therefore, a combined SAIL/ESTL test program is recommended.

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3.6.2 Palmdale, OV Ground Test

The Requirements/Parameter Description for the one-way doppler tracking measurement is TBD as listed in Reference F. This requirement should be specified.

3.7 Documentation Tree

Figure 1 shows the test plans and requirements documentation tree utilized for the C&T test survey.

3.8 Tests Schedule

Figure 2 presents a schedule of planned C&T system test periods for each major test location. Also shown are the major milestones which the MVP identifies as constraints against verification of VET requirements. Test schedules appear to be compatible with major program milestones. Schedule information was obtained from the Orbiter Project, Schedule and Status Summary, 3 February 1975.

4.0 CONCLUSIONS

This survey reveals the tests planned and the division of testing among the test locations for the Tracking Subsystem. The following is a listing of the changes recommended to these test plans and test locations.

- Add a Requirement/Parameter Description for the Ku Band Rendezvous Radar Long Range Mode to eliminate a TBD in Reference F (Palmdale TRSD).
- Delete the testing of Ku-Band rendezvous radar as listed in Reference G (ALT TRSD).
- KSC ground testing for the Rendezvous Radar/Comm A as listed in Reference N is too detailed and should be limited to self tests and system functional type tests.
- Revise Reference I OFT requirements document to include testing pertaining to the Rendezvous Data Acquisition and Convergence Times.

- Add the Requirement/Parameter Description for the one-way doppler tracking measurement to eliminate a TBD in Reference F (Palmdale TRSD).
- Add testing in the SAIL to accomplish verification of redundancy management. These tests should be added to the SAIL Integrated Test Plan, Reference Q.
- Revise the Reference J SAIL test requirements to show a combined SAIL/ESTL test program for Orbital Tracking test requirements.
- Update the VEI (Reference A) and the MVP (Reference B) as required to reflect use of C-Band beacon on OV 101.
- Delete the power output measurement of the TACAN air-to-air mode as listed in the ADI test document Reference M.
- Revise the Reference J SAIL test requirements document to include testing to verify systems operations of a TACAN unit with other vehicle subsystems.
- Add a Requirement/Parameter Description for the TACAN power output T/R mode to eliminate a TBD in Reference F (Palmdale TRSD).
- Add a Requirement/Parameter Description for C-Band Beacon functional verification to eliminate a TBD in Reference F (Palmdale TRSD).
- Change NAVAID measurements listed in Reference G (ALT TRSD) to specify RF power levels measured by using antenna couplers rather than using a hardline to the equipment.
- The SAIL should use flight type hardware for system verification testing rather than math models as specified in Reference K.
- Detailed testing such as MSBLS DME pulse width and interpulse spacing, coding rejection limits, AGC slow rates, spectrum check, pulse width, and repetition rates as specified in References G and R should be eliminated.

5.0 REFERENCE DOCUMENTS

- (A). Orbiter Vehicle End Item Specification For Space Shuttle System, Part I, MJ070-0001-1A, Change 5, December 2, 1974
- (B). Shuttle Master Verification Plan, Volume III, MJ072-0004-3, January 20, 1975 PDR Issue
- (C). Avionics Development Laboratory, Communications and Tracking, Phase I, Detailed Test Plan, SD74-SH-0306, February 3, 1975
- (D). Sail Test Requirements, MDC Working Paper, 1.3-WP-CO104-016, September 16, 1974
- (E). Space Shuttle Program, NASA Task 501, Communications and Tracking Systems Ground Testing, Program Plan, EE7-74-003, June 1974
- (F). Test Requirements Specification Document, In Process and Acceptance - Orbiter, MLO101-0001, Rev A. January 27, 1975
- (G). Orbiter Approach and Landing Test Program, Test Requirements and Specifications Document (TRSD), SD74-SH-0175, December 13, 1974, Second Coordination Draft
- (H). Orbiter Approach and Landing Test Requirements, JSC-08943, August 1, 1974, Baseline
- (I). Shuttle Orbital Flight Test Requirements, Draft, JSC-08576, January 15, 1975
- (J). SAIL Requirements Document, MDC Working Paper, 1.3-WP-CO102-022, October 29, 1974
- (K). Requirement/Definition Document, Book 12, Subsystem Ground Test (SAIL) SD72-SH-0112-20, Vol. 12-20, Rockwell International, January 21, 1974
- (L). SAIL Test Requirements, TRW letter 73: 7150:MF-42, November 2, 1973 (Review of Requirements/Definition Document, Avionics, Book 5, Section 4, SD-73-SH-105, July 2, 1973)
- (M). Avionics Development Lab, General Test Plan, SD73-SH-0298, no date, final
- (N). Launch Operations, Orbiter Tracking System, Test and Checkout Requirements, KSC, November 1, 1974, Rev. A, preliminary
- (O). Checkout Plan, Orbiter and Combined Elements Ground Operations, SD73-SH-0062, May 7, 1974

- (P). Space Shuttle Flight and Ground System Specification, JSC 07700
Vol X, Rev A, change 20
- (Q). Sail Integrated Test Plan Vol II, SD-74-SH-C215-2, December 20,
1974 (Rough Draft)
- (R). EAFB-ALT, Launch Operations, Orbiter Tracking System, Test and
Checkout Requirements, KSC, December 12, 1975, Preliminary
- (S). Launch Operations, Orbiter Communications System, Test and
Checkout Requirements, KSC, September 25, 1974, Rev A, Preliminary

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TABLE II - TEST LOCATION SUMMARY

VEI PARA. NO. (TITLE)	TEST LOCATIONS FOR TRACKING TESTS										COMMENTS
	ADL	SAIL	ESTL	SAIL/ ESTL	PALM- DALE	ALT		OFT			
						GND	FLI	GND	FLI		
3.3.5.2.2.7.1 Rendezvous Tracking General Performance		X			X	X ₁		X	(X)OV3	(X) Subcontractor Test Article (S/C TA)	
3.3.5.2.2.7.1.1 Rendezvous Tracking Acquisition Time		X									
3.3.5.2.2.7.2.1 Atmospheric Tracking	X				X	X		X	X	(X) S/C TA prime test area	
3.3.5.2.2.7.2.2 Approach and Landing Tracking	X	(X)			X	X	(X)	X		(X) S/C TA prime test area	
3.3.5.2.2.7.3 Orbital Tracking		X	(X)		X			X	(X)OV2		
<div>(X) Indicates prime test area, used for verification, as defined by the MVP. (Ref. B)</div> <div>X Indicates additional tests performed at this location.</div> <div>X₁ Test not required since there is no rendezvous radar on OV101</div>											

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TABLE III
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1	Rendezvous Tracking	1. Subcontractor Test Article 2. OFT-OV3
TEST LOCATION/TEST		REMARKS
<u>ADL</u> No tests indicated <u>(References C & M)</u> <u>SAIL</u> Verify ability of Rendezvous Radar to detect, acquire and automatically track passive or cooperative targets as specified for the purpose of determining range, range rate, angle and angle rate relative to the target. <u>(Reference K)</u> <u>ESTL</u> No tests indicated <u>(Reference E)</u> <u>SAIL/ESTL</u> No documentation available. <u>Palmdale, OV Ground Test</u> Ku-Band Radar/Comm Rendezvous Track. Verif. Radar Power on <ul style="list-style-type: none"> • Rendezvous Self Test • Rend Radar - Operate • Rend Radar - Ready • Rend Radar - Power On • Radar Temp • Radar RF Power • Radar Passive Mode Radar Search Mode <ul style="list-style-type: none"> • Radar Search • Target Tracking • Radar Detection • Scan Warn 		

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TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1	Rendezvous Tracking	1. Subcontractor Test Article 2. OFT-OV3
TEST LOCATION/TEST		REMARKS
<u>Palmdale, OV Ground Test (Continued)</u>		
Radar Manual Mode		
Radar Auto Mode		
<ul style="list-style-type: none"> • Auto Mode • Wide Width Scan 		
Radar Passive Mode		
<ul style="list-style-type: none"> • Radar Range Capability • Range Accuracy • Range Rate Capability • Range Rate Accuracy • Angle Capability • Angle Accuracy • Angle Rate Capability • Angle Rate Accuracy 		
Co-op Mode		
<ul style="list-style-type: none"> • Range • Range Accuracy • Range Rate • Range Rate Accuracy • Angle • Angle Accuracy • Angle Rate • Angle Rate Accuracy 		
Radar Long Range Mode (Reference F)		
<u>ALT (Ground)</u>		
Ku-Band Radar/Comm Rendezvous Tracking Verification (Reference G)		Delete reference to Ku-Band Rendezvous Tracking. No Rendezvous Radar on OV-101.
<u>ALT (Flight)</u>		
No tests indicated. (Reference H)		No Rendezvous Radar On OV- 101.

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TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1	Rendezvous Tracking	1. Subcontractor Test Article 2. OFT-OV3
TEST LOCATION/TEST		REMARKS
OFT (Ground)		
1. Rendezvous Radar/Comm A (OFF) <ul style="list-style-type: none"> a) Self Test b) Target track test <ul style="list-style-type: none"> • Co-op range & range rate check • Co-op angle & angle rate check • Co-op ready, search detect and tracking flag check • Co-op antenna scan check • Non co-op range & angle check • Co-op threshold c) Functional path delta <ul style="list-style-type: none"> • Slew test • Antenna display • Distance scale check • Angle scale check • Comm. A data check • Jettison circuit check • Heater temp. measurement • Standby mode check • Range gate designate d) Performance checks <ul style="list-style-type: none"> • Range limits co-op • Range limits passive • Range rate limit check • Pitch & roll angle coverage • Antenna drift rate • Transponder check e) RF checks <ul style="list-style-type: none"> • Transmit power • Minimum discernable signal • Pulse width, shape and rep rate checks • Transmit frequency & spectrum check • Receiver AGC check 		KSC ground testing is too detailed. Testing should be reduced to eliminate items such as testing black box specifications.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1	Rendezvous Tracking	1. Subcontractor Test Article 2. OFT-OV3
TEST LOCATION/TEST		REMARKS
<u>OFT (Ground) (Continued)</u>		
2. Comm B a) Self Test b) Target track test • Lock on, angle check & Tracking • Auto search & detect check • Manual antenna search & detect check • Threshold c) Function path delta • Antenna deploy • Slew test • Comm B data check • Jettison circuit check • Heater temp. measurement • Standby mode check d) Antenna performance check • Pitch & roll angle • Antenna drift rate e) RF checks • Transmit power • Minimum discernable signal • Transmit frequency & spectrum check • Receiver AGC check <u>(Reference N)</u>		
<u>OFT (Flight)</u>		
1. Rendezvous Radar Performance Test Verify rendezvous radar, angle rate, range and range rate accuracy performance is within specified on-orbit performance limits required for efficient rendezvous operations.		Add angle accuracy test.
2. Orbital Rendezvous Objective: To demonstrate closed loop performance of GNC system in performing orbital rendezvous using both the star tracker and rendezvous radar. <u>(Reference I)</u>		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1.1	Rendezvous Data Acquisition and Convergence Time	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>ADL</u> No tests indicated <u>(References C&M)</u> <u>SAIL</u> Verify ability to acquire/reacquire a rendezvous target via radar within the time limits specified. <u>(Reference K)</u> <u>ESTL</u> No tests indicated <u>(Reference E)</u> <u>SAIL/ESTL</u> No documentation available <u>Palmdale, OV Ground Test</u> No tests indicated <u>(Reference F)</u> <u>ALT (Ground)</u> No tests indicated <u>(Reference G)</u> <u>ALT (Flight)</u> No tests indicated. <u>(Reference H)</u> <u>OFT (Ground)</u> No tests indicated. <u>(Reference N)</u>		Include Acquisition & Convergence Time testing as performed in SAIL.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.1.1	Rendezvous Data Acquisition and Convergence times	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>OFT (Flight)</u> No test indicated. <u>(Reference I)</u>		Add Rendezvous Data Acquisition and Convergence Times testing to OFT flight test program. This test could be part of the rendezvous radar general performance test.

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TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>ADL</u> <u>TACAN</u> 1. Initial testing of the TACAN subsystem will verify the accuracy, validity, and drift trends of all functional outputs with the use of ground station simulators, special test equipment, and general laboratory test equipment. Development and trend testing of the operational range will include power output of transmit/receive (T/R) and air-to-air (A/A) modes, and receiver sensitivity. Other parameters to be verified, and trends developed are: bearing accuracy, distance accuracy, selectivity and select time, response speed (distance and bearing), tracking rate, and memory capability. Digital output compatibility with special MDM input/output circuitry will also be developed. <u>(Reference M)</u> 2. <u>Serial Digital</u> <ul style="list-style-type: none"> Distance and Bearing Data Output Verify the T/R TACAN transmits to the MDM, 32 bit ternary serial digital words which shall contain distance and bearing information, derived from the STE. This data output will be correlated with the software. Serial Digital Data Channel Signal Characteristics 		A/A mode will not be implemented.

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TABLE III (CONT'D)
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VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>ADL (continued)</u> 3. Transmission Method • Data Code • Data Rate • Clock • Word Sync • Word Size and Format 4. Analog Signals • Beacon Identify Tone 5. Input/Output Singals • Blanking Pulse • Radio Frequency <u>(Reference C)</u> <u>SAIL</u> No tests indicated. <u>(Reference J)</u> <u>ESTL</u> No tests indicated. <u>(Reference E)</u> <u>SAIL/ESTL</u> No documentation available.		Should perform a test utilizing a TACAN unit to verify system operations including redundancy management.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>Palmdale, CV Ground Test</u> TACAN Functional Verification TACAN Activation & Warmup Test TACAN 1, 2, 3 Power On TACAN Verification TACAN 1, 2, 3 Tests o Range o Bearing Operational Range TACAN 1, 2, 3 Range & Range Flag Operational Bearing TACAN 1, 2, 3 Bearing & Bearing Flag Accuracy, 15 Hz Digital Bearing TACAN 1, 2, 3 Distance Tracking Rate Verif TACAN 1, 2, 3 Bearing Tracking Rate Verif TACAN 1, 2, 3 Response Speed Verification TACAN 1, 2, 3 DME Mode & Bearing Mode TACAN X-Y Mode Select Test TACAN 1, 2, 3 Channel Select Time Test TACAN 1, 2, 3 (Auto Mode) Receive Only Mode Test TACAN 1, 2, 3 No RF output. TACAN Memory Verif. TACAN 1, 2, 3 Distance Memory & Bearing Memory TACAN Transmitter Frequency Freq. 1, 2, 3 X & Y Mode Verif		

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TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>Palmdale, OV Ground Test (Continued)</u> Blanking Test TACAN 1, 2, 3 Antenna Switching Auto mode with no signal input TACAN 1, 2, 3 Ant switching rate Manual Mode - Verif RF signal from selected antenna TACAN power output, T/R mode TACAN 1, 2, 3 C-Band Beacon Functional Verification (Reference F) <u>ALT (Ground)</u> 1. TACAN a) Self Test b) Functional path delta • Channel select X and Y mode • Threshold • Receive only mode • Antenna switching • Blanking Test • Power measurement c) Performance Check • Memory Check • Range limits • Bearing limits • Adjacent channel rejection • Dynamic tracking • Ground station check d) RF Checks • Minimum discernable signal • Transmit frequency & spectrum check • Transmit pulse width & rep rate		The test method for TACAN power output, T/R mode is TBD. See para. 3.4.3 of this report. C-Band Beacon Functional Verification tests need to be determined (use tests suggested by FRC/EAFFB). Many of these test require- ments are duplicates or more detailed than those for Palmdale. It is recommended that performance checks such as memory check, range and bearing limits, and adjacent channel rejection be deleted.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>ALT (Ground) (Continued)</u> m. TACAN Transmitter Freq. TACAN 1, 2 & 3 X & Y Mode Verif. n. Blanking Test TACAN 1, 2 & 3 o. Antenna Switching • Auto Mode TACAN 1, 2 & 3 Antenna Switching Rate • Manual Mode 4. C-Band Beacon (OV 101 only) Functional Verification <u>(Reference G)</u> <u>ALT (Flight)</u> TACAN Objectives: 1. Verify TACAN range and bearing accuracy performance. 2. Verify adequacy of display manual approaches. <u>(Reference H)</u> <u>CFT (Ground)</u> 1. TACAN (at OFF) a) Self Test b) Functional path delta • Channel select X & Y mode • Threshold • Receive only mode • Antenna switching • Blanking test • Power measurement c) Performance check • Memory check • Range limits • Bearing limits • Adjacent channel rejection		
		CFT Ground tests appear too detailed. See paragraphs 3.4.5.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.1	Atmospheric Flight Support Tracking	Subcontractor Test Article
TEST LOCATION/TEST		REMARKS
<u>OFT (Ground (Continued))</u> <ul style="list-style-type: none"> ● Dynamic Tracking ● Ground station check d) RF checks <ul style="list-style-type: none"> ● Minimum discernable signal ● Transmit frequency & spectrum check ● Transmit pulse width & rep rate 2. TACAN (at PAD) <ul style="list-style-type: none"> a) Self test b) Functional path delta <ul style="list-style-type: none"> ● TACAN beacon check ● Channel select ● Threshold ● Receive only mode ● Blanking test ● Power measurement <u>OFT (Flight)</u> <p>High Altitude TACAN Performance Objectives: Verify the acquisition constraints and TACAN performance after the Orbiter exits blackout. Determine L-Band blackout constraints. <u>(Reference I)</u></p>		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEL PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Tracking	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<p><u>ADL</u></p> <p>1. Microwave Scan Beam Landing System</p> <p>The DME transmitter power, center frequency, and stability trends will be investigated. Receiver sensitivity and center frequency will be used to verify adjacent channel rejection, channelization, and compatibility of the digital information output to the MDM. Ground station simulators will provide a cooperative station for landing system mode verification.</p> <p>2. Radar Altimeter</p> <p>The radar altimeter compatibility with the special MDM input/output module will be analyzed. Altitude accuracy, tracking rate, maximum acquisition range, and accuracy at all bank angles of an orbiter landing situation will be verified. (Reference M)</p> <p>3. MSBLS</p> <ul style="list-style-type: none"> • Power on tests • Signal characteristics • Transmission tests • Output tests • Operation • Accuracy • Frequency-channel tests • Guidance tests • Dynamic Range and coverage tests • Sensitivity • Built-in test <p>4. Radar Altimeter</p> <ul style="list-style-type: none"> • Prerequisite measurements • Power up tests 		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Tracking	1. SAIL 2. CV-101
TEST LOCATION/TEST		REMARKS
<u>ADL (Continued)</u> <ul style="list-style-type: none"> ● Verify timing, waveform, and voltage amplitude of MDM ● Verify analog altitude from RA ● Verify analog and digital altitude outputs identical within allowable tolerance. ● Perform bit comparison test of digital altitude data outputs (from RA, from MDM) ● Determine RA altitude accuracy using fixed program pulse delays corresponding to altitude between 0-2500 ft. ● Verify that RA will maintain reliable track at altitude rate of +/-2000 ft/sec ● Determine the degradation of BER resulting from the injection of TBD specified magnitudes of noise at specific points. <u>(Reference C)</u>		
<u>SAIL</u> Verify ability to obtain position information relative to the runway during approach and landing to support autoland with range, coverage and accuracy specified in VEI table 3.3.5.2.2.7.2. Evaluate using math models of the landing aid RF equipment <u>(Reference K)</u>		Should verify system by utilizing NAVAID hardware (USBL & RA) rather than using math models.
<u>ESTL</u> No tests indicated. <u>(Reference E)</u>		
<u>SAIL/ESTL</u> No documentation available.		

TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.7.2.7.2.2	Final Approach and Landing Tracking	SAIL
TEST LOCATION/TEST		REMARKS
<p><u>Palmdale, OV Ground Test</u></p> <p>1. Radar Altimeter Functional Verification Activation</p> <p>Radar ALT No. 1, 2-on Radar ALT No. 1, 2-Flag</p> <p>Altimeter Verification</p> <p>Radar ALT. No. 1, 2-Test (Stimuli on) Radar ALT. No. 1, 2 Altitude Radar ALT. No. 1, 2 Operating Freq. Radar ALT. No. 1, 2 Power Output Radar ALT. No. 1, 2 Pulse Width Radar ALT. No. 1, 2 Sensitivity Radar ALT. System delay (each system) Radar ALT. No. 1, 2 Range ALT. No. 1, 2 PRF ALT. No. 1, 2 Altitude Above Floor</p> <p>2. MSBLS Functional Verification</p> <p>MSBLS Activation</p> <p>MSBLS 1, 2, 3 Power</p> <p>MSBLS Self Test MSBLS 1, 2, 3:</p> <ul style="list-style-type: none"> • Azimuth flag • Range flag • Elevation flag • Azimuth • Range • Elevation <p>Elevation Guidance Test MSBLS 1, 2, 3 Elev Flag & Output</p>		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Tracking	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<u>Palmdale, OV Ground Test (Continued)</u> Azimuth Guidance Test MSBLS 1, 2, 3, Az Flag & Output Range Information (DME) test MSBLS 1, 2, 3 Range Flag & Output MSBLS 1,2,3 RCVR/XMTR Channel Verif RCVR Channels 1 through 10 XMTR center frequency (DME) Adjacent Channel Rejection MSBLS 1, 2, 3 Digital Interface-MSBLS 1, 2, 3 Parity Velocity DME Pulse Parameter Test-MSBLS 1, 2, 3 Pulse width Interpulse spacing MSBLS RF Power Output-DME MSBLS 1, 2 & 3 <u>(Reference F)</u> <u>ALT (Ground)</u> 1. MSBLS a) Self test b) Functional path delta • Channel select • Threshold • DME power c) Performance check • Elevation limits • Azimuth limits • Range limits • Coding rejection limits* • Angle rate limits*		ALT Ground tests appear too detailed. Tests such as coding rejection limits, AGC slew rate, spectrum checks, pulse width, and rep rates need not be tested for ALT. Consideration should be made for tests of a self test nature only; especially for turnaround testing.

* Planned to be deleted

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Tracking	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<u>ALT (Ground) (Continued)</u> d) RF checks <ul style="list-style-type: none"> • Minimum discernible signal • DME pulse width & rep rate • DME frequency & spectrum check • AGC slew rate check* 2. Radar Altimeter <ul style="list-style-type: none"> a) Self Test b) Altitude above ground c) Functional path delta <ul style="list-style-type: none"> • Threshold • Power d) Performance check <ul style="list-style-type: none"> • Altitude limits • System delay e) RF checks <ul style="list-style-type: none"> • Minimum discernable signal • Transmit pulse width & rep rate • Transmit frequency & spectrum check (Reference R)		
3. Radar Altimeter Functional Verif. <ul style="list-style-type: none"> a. Activation RA 1, 2 - on & flag b. Altitude verif. RA 1, 2 - Test & alt. RA 1, 2 - Alt. above floor 		Power output measurements (out of antenna) should be made for the Radar Altimeter. See para. 3.5.2 of this paper
4. MSBLS Functional Verif. <ul style="list-style-type: none"> a. MSBLS Activation MSBLS 1, 2 & 3 - power on b. MSBLS Self test MSBLS 1, 2 & 3 - Test, azimuth flag, range flag & elevation flag c. Elevation Guidance Test MSBLS 1, 2, 3 - Output & Elev flag d. Azimuth Guidance Test MSBLS 1, 2 & 3 - Output & azimuth flag 		Power output measurements (out of antenna) should be made for the MSBLS. See para. 3.5.2 of this paper.

* Planned to be deleted.

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VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7-2.2	Final Approach and Landing Tracking	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<p><u>ALT (Ground) (Continued)</u></p> <ul style="list-style-type: none"> e. Range info (DME) test MSBLS 1, 2 & 3 - Output & range flag f. MSBLS 1, 2 & 3 RCVR/XMTR Channel verif. RCVR channels 1 through 10 XMTR center frequency (DME) g. MSBLS 1, 2 & 3 Antenna isolation h. MSBLS 1, 2 & 3 - AGC dynamic range i. MSBLS 1, 2 & 3 Digital interface <ul style="list-style-type: none"> • parity • validity j. MSBLS 1, 2 & 3 DME Pulse parameter test <ul style="list-style-type: none"> • Pulse width • Interpulse spacing <p><u>(Reference G)</u></p> <p><u>ALT (Flight)</u></p> <p>1. MLS Performance OBJECTIVE:</p> <p>Verify the performance of the Microwave Scanning Beam Landing System during the final approach and landing phase. The capability to provide display data required for manual instrument landing and the capability for autoland through rollout shall be verified.</p> <p>2. Radar Altimeter Performance OBJECTIVE:</p> <p>Verify the altitude capability and accuracy of the radar altimeter. <u>(Reference H)</u></p>		<p>These tests (g through j) are too detailed for ground operations testing. See para. 3.5.2 of this paper.</p>

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Test	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<u>OFT (Ground)</u> 1. MSBLS (at OFF) a) Self test b) Functional path delta <ul style="list-style-type: none"> • Channel select • Threshold • DME power c) Performance check <ul style="list-style-type: none"> • Elevation limits • Azimuth limits • Range limits • Coding rejection limits • Range rate limits • Angle rate limits d) RF checks <ul style="list-style-type: none"> • Minimum discernible signal • DME pulse width & rep rate • DME frequency & spectrum check • AGC slew rate check 2. MSBLS (at PAD) a) Self test b) Functional path delta <ul style="list-style-type: none"> • Channel select • DME power • Threshold 3. Radar Altimeter (at OFF) a) Self test b) Altitude above ground c) Functional path delta <ul style="list-style-type: none"> • Threshold • Power d) Performance check <ul style="list-style-type: none"> • Altitude limits • System delay e) RF checks <ul style="list-style-type: none"> • Minimum discernible signal • Transmit pulse width & rep rate • Transmit frequency & spectrum check 		KSC ground testing is too detailed. Tests such as DME pulse width and rep rate, spectrum checks and AGC slew rate should be deleted.

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.2.2	Final Approach and Landing Test	1. SAIL 2. OV-101
TEST LOCATION/TEST		REMARKS
<p><u>OFT (Ground) (Continued)</u></p> <p>4. Radar Altimeter (at PAD)</p> <p> a) Self test</p> <p> b) Altitude ranging off ET</p> <p><u>(Reference N)</u></p> <p><u>OFT (Flight)</u></p> <p>No tests indicated.</p> <p><u>(Reference I)</u></p>		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.3	Orbital Tracking	1. ESTL 2. OFT-OV2
TEST LOCATION/TEST		REMARKS
<p><u>ADL</u></p> <p>No tests indicated. (References C & M)</p> <p><u>SAIL</u></p> <p>1. Verify ability to provide coherent turnaround of the S-Band carrier to permit tracking by STDN, TDRS and AFSCF and to extract one-way doppler from STDN, TDRS or AFSCF S-Band signals. Accuracy and stability shall be specified. (Reference K)</p> <p>2. Verify orbiter STDN tracking provisions are provided in accordance with the STDN link. Conduct laboratory test of C&T STDN tracking provisions to establish compliance with performance requirements.</p> <p>3. Verify ability to support STDN-Direct tracking by turnaround of carrier phase to permit range rate determination by STDN ground stations Verify using simulated STDN tracking link.</p> <p>4. Verify orbiter ability to support AFSCF tracking by turnaround of carrier phase to permit range rate determination by the ground stations. Verify using simulated STDN tracking links.</p> <p>5. Verify AFSCF tracking provisions (on-board) are provided in accordance with the AFSCF link. Conduct laboratory tests of C&T AFSCF tracking provisions to verify compliance with performance requirements. (Reference L)</p>		

TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.3	Orbital Tracking	1. ESTL 2. OFT-OV2
TEST LOCATION/TEST		REMARKS
<u>ESTL</u> 1. Verify STDN tracking provisions are provided in accordance with STDN link. 2. Verify TDRS tracking provisions are provided in accordance with the TDRS link 3. Verify orbiter ability to support AFSCF tracking by turnaround of carrier phase to permit range rate determination by the ground station 4. Verify ability to support STDN-Direct tracking by turnaround of carrier phase to permit range rate determination by STDN ground station. 5. Perform acquisition test on all subsystems involved in frequency acquisition (with and without doppler): • Transponder frequency tracking capability • Receiver frequency tracking capability • Acquisition times • Probability of first sweep acquisition. <u>(Reference E)</u> <u>SAIL/ESTL</u> No documentation available. <u>Palmale, OV Ground Test</u> S-Band Verification - STDN, SGLS FM Xponder Carrier Doppler Tracking Acquisition modes Acquisition - with doppler 2 Way Doppler tracking MSMT No. 1 Orbiter frequency no. 1		These test requirements are nearly the same as the Orbital Tracking test requirements for SAIL listed above. Therefore, a combined ESTL/SAIL test program is recommended. (See para. 3.6 of this paper.)

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TABLE III (CONT'D)
TEST REQUIREMENT SUMMARY

VEL PARA. NO.	REQUIREMENT SUMMARY	MVP VERIFICATION PROGRAM
3.3.5.2.2.7.3	Orbital Tracking	ESTL OV2
TEST LOCATION/TEST		REMARKS
<u>Palmdale, OV Ground Test (Continued)</u>		
Orbiter frequency No. 2 Receiver AGC Receiver static phase error		
2 Way Doppler tracking MSMT No. 2		
Orbiter frequency No. 1 Orbiter frequency No. 2 Receiver AGC Receiver static phase error		
2 Way doppler tracking MSMT No. 3		
Orbiter frequency No. 1 Orbiter frequency No. 2 Receiver AGC Receiver static phase error		
One-Way doppler tracking MSMT <u>(Reference F)</u>		One-way doppler tracking MSMT's Requirements/ Parameter Description are TBD.
<u>ALT (Ground)</u>		System not installed on OV101
No tests indicated <u>(Reference G)</u>		
<u>ALT (Flight)</u>		
No tests indicated <u>(Reference H)</u>		
<u>GFT (Ground)</u>		
1. S-Band Xponder System Threshold • Doppler ranging		

TABLE III (COMPLETED)
TEST REQUIREMENT SUMMARY

VEI PARA. NO.	REQUIREMENT SUMMARY	MVF VERIFICATION PROGRAM
3.3.5.2.2.7.3	Orbital Tracking	1. ESTL 2. OFT-OV2
TEST LOCATION/TEST		REMARKS
<u>OFT (Ground) (Continued)</u> 2. S-Band Xponder System Delay (Orbiter system turnaround delay) • Doppler ranging STDN SGLS TDRS 3. S-Band Xponder RCV Only (STDN, SGLS, TDRS) Verify operation of RCV only made with coherent signal and verify uplink data and one-way doppler extraction capability. (Reference 3)		
<u>OFT (Flight)</u> 1. On Orbit Navigation Performance Objective: Verify capability to determine and maintain orbiter position within specified accuracies during all on-orbit operating modes utilizing on-orbit one way (onboard) and two way (ground system) doppler navigation (Reference 1)		

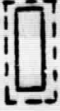
INDICATES DOCUMENT IS NOT
REFERENCED IN THIS REPORT

COORDINATION DRAFT

RELEASED

FINAL

PRELIMINARY



JSC 07700
VOL X SYS SPEC
REV A, CHANGE 20

SMVP - VOL I
GENERAL JSC-07700
10-NVP-01 REV A
CHANGE 2

SMVP-VOL III
ORBITER
MJ072-0004-3

ORBITER VEL
PART 1
MJ070-0001-1A
CHANGE 5

KSC GROUND
TEST REQ
REV A

AVIONICS
DEVL TEST &
SIM PLAN
SD73-SH-0133

ORB & COMB
ELEM C/O PLAN
SD73-SH-0062

ORBITER ALT
TRSD (GND)
SD74-SH-0175

IN-PROCESS
ACCEPTANCE TRSD
MD101-0001

ADL GEN TEST
PLAN
SD73-SH-0298

SAIL RDD BOOK 12
SD72-SH-0112-20
VOL 12-20

ORBITER ALT
TEST REQ (FLT)
JSC-08943

SHUTTLE ORB
FLT TEST REQ
JSC-03576

NASA TASK 501
C&T SYS CHD
TESTING
EE7-74-003

C&T PHASE I
DETAILED TEST
PLAN
SD74-SH-0306

SAIL INTR
TEST PLAN
SD74-SH-0215-2
-2

ALT C/O
REQ. KSC

FIGURE 1

APPLICABLE ORBITER TRACKING TEST
PLANS AND REQUIREMENTS DOCUMENTATION

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FIGURE 2 C&T TEST SCHEDULE

